

# Multiplication Table

Levi-Civita

1	i
i	-1

$$(u_1 + iu_2)^2 = u_1^2 - u_2^2 + i(2u_1u_2)$$

Transformation Matrix

$$\begin{pmatrix} u_1 - u_2 \\ +u_2 \ u_1 \end{pmatrix}$$

$$\begin{aligned} L(u^T) \cdot \vec{u} &= \vec{v} \\ 2L(u^T)u &= \vec{v} \end{aligned}$$

(1)

1	i	j	k
i	-1	k	-j
j	-k	-1	i
k	j	i	-1

$$-1 = i^2 = j^2 = k^2 = ijk$$

$$(u_1 + iu_2 + ju_3 + ku_4)^2$$

$L(u)$ , with  $L(u) \cdot u = u^2$

Quaternion

$$\begin{pmatrix} u_1 - u_2 - u_3 - u_4 \\ u_2 \ u_1 \ u_4 \ -u_3 \\ u_3 \ -u_4 \ u_1 \ u_2 \\ u_4 \ u_3 \ -u_2 \ u_1 \end{pmatrix} \begin{pmatrix} u_1 \\ u_2 \\ u_3 \\ u_4 \end{pmatrix} = \begin{pmatrix} u_1^2 - u_2^2 - u_3^2 - u_4^2 \\ 2u_1u_2 \\ 2u_1u_3 \\ 2u_1u_4 \end{pmatrix}$$

# Multiplication Table

(2)

1	i	j	k
1	i	j	-k
i	-1	k	j
j	-k	-1	-i
k	j	-i	1

$$k^2 = +1!$$

Koordinatenvektor Steifel

$$\begin{pmatrix} u_1^2 - u_2^2 - u_3^2 + u_4^2 \\ 2u_1u_2 - 2u_3u_4 \\ 2u_1u_3 + 2u_2u_4 \\ 0 \end{pmatrix} = \begin{pmatrix} u_1 & -u_2 & -u_3 & u_4 \\ u_2 & u_1 & -u_4 & -u_3 \\ u_3 & u_4 & u_1 & u_2 \\ u_4 & -u_3 & u_2 & -u_1 \end{pmatrix} \begin{pmatrix} u_1 \\ u_2 \\ u_3 \\ u_4 \end{pmatrix}$$



# 1. TWO-BODY PROBLEM

(1) (10)

Hamiltonian  $H = T + V$  of 2-b-Pr.

$$H = \frac{p_1^2}{2m_1} + \frac{p_2^2}{2m_2} - \frac{Gm_1m_2}{r} = H_{\text{cm}} + H_{\text{rel}}$$

$$\vec{p}_1 = m_1 \vec{v}_1 \quad ; \quad \vec{p}_2 = m_2 \vec{v}_2 \quad ; \quad r = |\vec{r}|$$

Move to relative coordinates

$$\begin{aligned} \vec{v} &= \vec{v}_1 - \vec{v}_2 \quad ; \quad \vec{r} = \vec{r}_1 - \vec{r}_2 \quad ; \quad \vec{p} = M \vec{v} \\ \mu &= \frac{m_1 m_2}{m_1 + m_2} \quad ; \quad M = m_1 + m_2 \quad ; \quad \vec{r}_{\text{cm}} = \frac{m_1 \vec{r}_1 + m_2 \vec{r}_2}{m_1 + m_2} \\ \vec{v}_{\text{cm}} &= \frac{m_1 \vec{v}_1 + m_2 \vec{v}_2}{m_1 + m_2} \quad ; \quad \vec{p}_{\text{cm}} = M \vec{v}_{\text{cm}} = \vec{p}_1 + \vec{p}_2 \end{aligned}$$

$$H_{\text{cm}} = \frac{p_{\text{cm}}^2}{2M} \quad H_{\text{rel}} = \frac{p^2}{2\mu} - \frac{Gm_1m_2}{r}$$

Canonical Equations:

$$\frac{\partial}{\partial \vec{p}_{\text{cm}}} H_{\text{cm}} = - \frac{\partial H_{\text{cm}}}{\partial \vec{r}_{\text{cm}}} = 0 \Rightarrow \vec{v}_{\text{cm}} = \text{const. (3 const.)}$$

$$\frac{\partial}{\partial \vec{r}_{\text{cm}}} H_{\text{cm}} = \frac{\partial H_{\text{cm}}}{\partial \vec{p}_{\text{cm}}} = \vec{p}_{\text{cm}} = M \vec{v}_{\text{cm}} \quad (\text{Def of } \vec{p}_{\text{cm}})$$

Conservation law:  $\vec{r}_{\text{cm}} \times \vec{v}_{\text{cm}} = \text{const.}$

$$\begin{matrix} \vec{v}_{\text{cm} 1, 2, 3} \\ \begin{matrix} r_1 v_2 - r_2 v_1 & r_3 v_1 - r_1 v_3 \\ r_2 v_3 - r_3 v_2 \end{matrix} \end{matrix}$$



# Relative Motion

$$(i) \quad \vec{p} = - \frac{\partial H}{\partial \vec{r}} = - \frac{Gm\mu}{r^2} \frac{\vec{r}}{r} = \mu \vec{v}$$

$$\Rightarrow \quad \vec{v} = - \frac{GM}{r^2} \frac{\vec{r}}{r}$$

$$(ii) \quad \vec{r} = \frac{\partial H}{\partial \vec{p}} = \frac{\vec{p}}{\mu} = \vec{v}$$

$$H_{rel} = \frac{p^2}{2\mu} - \frac{Gm\mu}{r}$$



(3)

Some conserved quantities:

$$\bullet E = \frac{1}{2} \mu v^2 - \frac{G M m \mu}{r} \quad ; \quad \frac{E}{\mu} = \frac{1}{2} v^2 - \frac{GM}{r}$$

$$\bullet \text{ Specific angular momentum: } \vec{j} = \vec{r} \times \vec{v}$$

$$\frac{d}{dt} \vec{j} = \frac{d}{dt} (\vec{r} \times \vec{v}) = \vec{v} \times \vec{v} + \vec{r} \times \dot{\vec{v}} = -\frac{GM}{r^3} \vec{r} \times \vec{v} = 0$$

$\bullet$  Runge-Lenz (eccentricity vector)

$$\vec{e} = \frac{\vec{v} \times \vec{j}}{GM} - \frac{\vec{r}}{r} \quad \text{is constant of motion and } |\vec{e}| = e \text{ ecc.}$$

(Note: not all of  $E, \vec{j}, \vec{e}$  are isolating!  
not all are even independent!)

Proof:

$$\frac{d\vec{e}}{dt} \stackrel{?}{=} -\frac{\vec{r} \times \vec{j}}{r^3} \rightarrow \frac{d}{dt} \left( \frac{\vec{r}}{r} \right) = \frac{\vec{j} \times \vec{r}}{r^3} - \frac{d}{dt} \left( \frac{\vec{r}}{r} \right)$$

$$\frac{\vec{j} \times \vec{r}}{r^3} = \frac{(\vec{r} \times \vec{v}) \times \vec{r}}{r^3} = \frac{\vec{v}(r^2 - \vec{r}(\vec{v} \cdot \vec{r}))}{r^3}$$

$$\begin{aligned} (A \times B) \times C &= B(C \cdot A) - A(B \cdot C) \\ &= \frac{\vec{v}}{r} - (\vec{v} \cdot \vec{r}) \frac{\vec{r}}{r^3} \stackrel{!}{=} \frac{d}{dt} \left( \frac{\vec{r}}{r} \right) \end{aligned}$$



$$\frac{d}{dt} \frac{\vec{r}}{r} = \frac{\vec{v}}{r} - \frac{\vec{r}}{r^2} \cdot \frac{dr}{dt} \quad r = \sqrt{x^2 + y^2 + z^2} \quad 3a$$

$$\frac{d}{dt} \frac{x}{r} = \frac{v_x}{r} - \frac{x}{r^2} \cdot \frac{dr}{dt}$$

$$= \frac{v_x}{r} - \frac{x}{r^2} \frac{d(x \cdot v_x + y \cdot v_y + z \cdot v_z)}{dt} \quad \checkmark$$

Isolating Integrals:

- generally only  $E$
- sph. symm. pot.  $E, \vec{J}$
- axi symm. pot.  $E, \vec{J}_z, [I_3]$
- 2 body problem  $E, \vec{J}$
- N body



Something more:

(4)

$$\vec{e} \cdot \vec{r} + r = \frac{\vec{r} \cdot (\vec{v} \times \vec{j})}{GM} = \frac{(\vec{r} \times \vec{v}) \cdot \vec{j}}{GM} = \frac{J^2}{GM}$$

$$r + \vec{e} \cdot \vec{r} = e r \cos \theta + r = r(1 + e \cos \theta)$$

$$r = \frac{J^2/GM}{(1 + e \cos \theta)} \quad \text{conic section}$$

$$r_{\min} = \frac{J^2/GM}{(1+e)} \quad ; \quad r_{\max} = \frac{J^2/GM}{(1-e)}$$

~~$a = \frac{r_{\min} + r_{\max}}{2} = \frac{J^2}{2GM} \left( \frac{1}{1+e} + \frac{1}{1-e} \right)$~~

$$a = (r_{\min} + r_{\max})/2 = \frac{J^2}{2GM} \left( \frac{1}{1+e} + \frac{1}{1-e} \right)$$

$$= \frac{J^2}{GM(1-e^2)} \Rightarrow J^2 = GMa(1-e^2)$$

$$r_{\min} = a(1-e)$$

$$r_{\max} = a(1+e) \rightarrow F$$



Energy, e.g. at pericenter:

$$\frac{E}{\mu} = \frac{1}{2} v^2 - \frac{GM}{r} = \frac{1}{2} \frac{j^2}{r_{\min}^2} - \frac{GM}{r_{\min}}$$

$$= \frac{GMa(1-e^2)}{2a^2(1-e)^2} - \frac{GM}{a(1-e)}$$

$$= \frac{GM}{2a} \left( \frac{1+e}{1-e} - \frac{2}{1-e} \right)$$

$$\frac{-1+e}{1-e} = - \frac{1-e}{1-e}$$

$$\frac{E}{\mu} = - \frac{GM}{2a}$$

$$E = - \frac{Gm_1 m_2}{2a}$$



(48)

At  $r_{min}$ : (Show that  $\vec{e}$  points to pericenter)

$$\vec{e} \cdot \vec{r} = \frac{z^2}{GM} - r_{min}$$
$$= \frac{z^2}{GM} - \frac{z^2}{GM(1+e)}$$

$$= \frac{z^2}{GM} \left( 1 - \frac{1}{1+e} \right)$$
$$\frac{ze}{1+e}$$

~~Answer~~

$$\left( \vec{e} \cdot \frac{\vec{r}_{min}}{r_{min}} \right) \cdot \frac{GM}{GM(1+e)}$$

$$\vec{e} \cdot \frac{\vec{r}_{min}}{r_{min}} = e$$

$$|\vec{e}| \left| \frac{\vec{r}_{min}}{r} \right| \cos(\vec{e}, \vec{r}_{min}) = e \text{ — angle } 0!$$



Relation to classical orbital elements: (5)

We have  $E, \vec{j}$ , Pos. of Pericenter. (5)

(Note  $\vec{e}, \vec{j}$  not independent:  $\vec{e} \cdot \vec{j} = 0$ )

$$i = \arccos \frac{j_z}{j} \quad (\Leftrightarrow) \quad \cos i = \frac{j_z}{j}$$

Inclination

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$$\vec{n} = \vec{e}_z \times \vec{j} = (-j_y, j_x, 0) = (n_x, n_y, 0)$$

$$\Omega = \arccos \frac{n_x}{|\vec{n}|} \quad (\Leftrightarrow) \quad \cos \Omega = \frac{n_x}{|\vec{n}|}$$

$$n_y \geq 0$$

$$2\pi - \Omega = \arccos \frac{n_x}{|\vec{n}|} \quad (\Leftrightarrow) \quad \cos(2\pi - \Omega) = \frac{n_x}{|\vec{n}|}$$

Longitude of  
Ascending Node

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$$n_y < 0$$

$$\omega = \arccos \frac{\vec{n} \cdot \vec{e}}{|\vec{n}|e} \quad (\Leftrightarrow) \quad \cos \omega = \frac{\vec{n} \cdot \vec{e}}{|\vec{n}|e}$$

Argument of Periapsis